

Norm Ovens Rockwell Collins

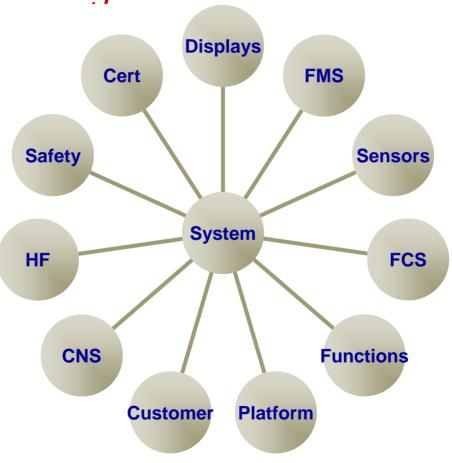
2005 Software and Complex Electronic Hardware Standardization Conference





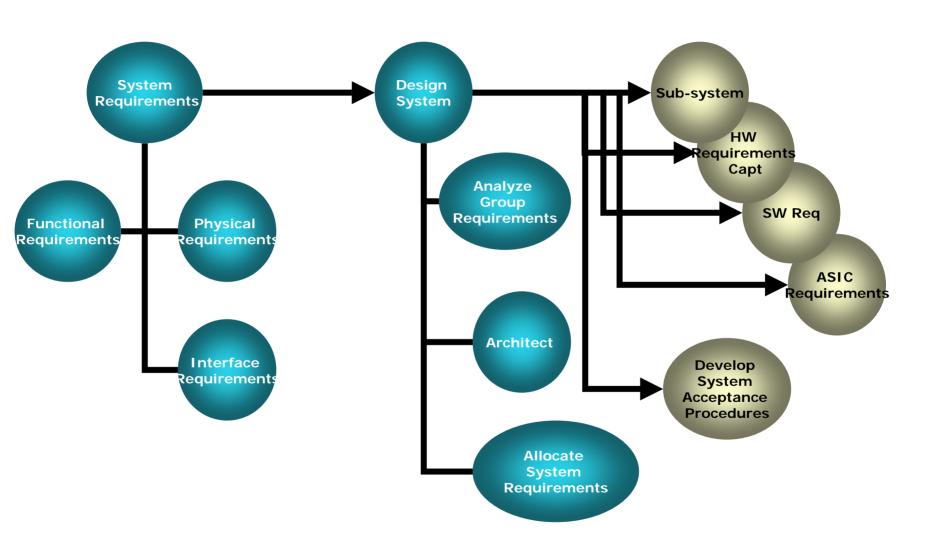
- System design
- Reuse
- Integrated modular avionics
- Certification





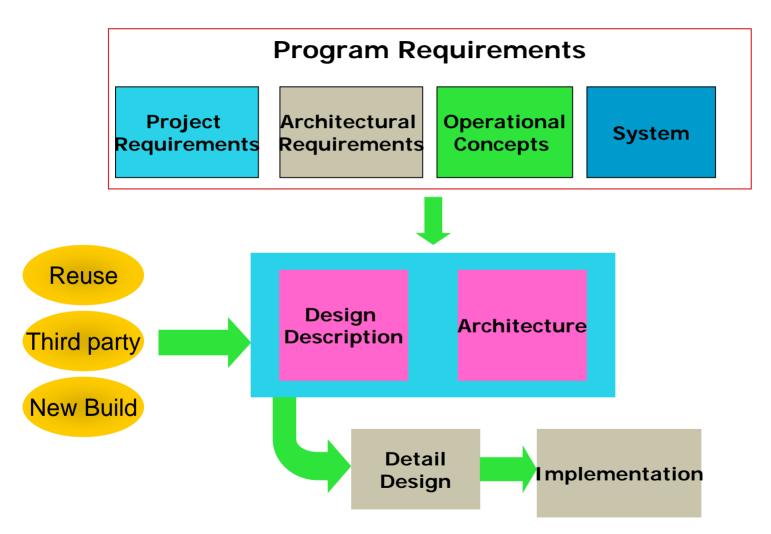


Technical Consistent Process(TCP) Design System





System Development Process





System Design Goals

- Customer requirements
 - Functionality
 - Design for reliability
 - Design for dispatch
- Design for safety
 - Availability
 - Integrity

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System Design Goals

Design to minimize:

- Cost
- Size, weight and power

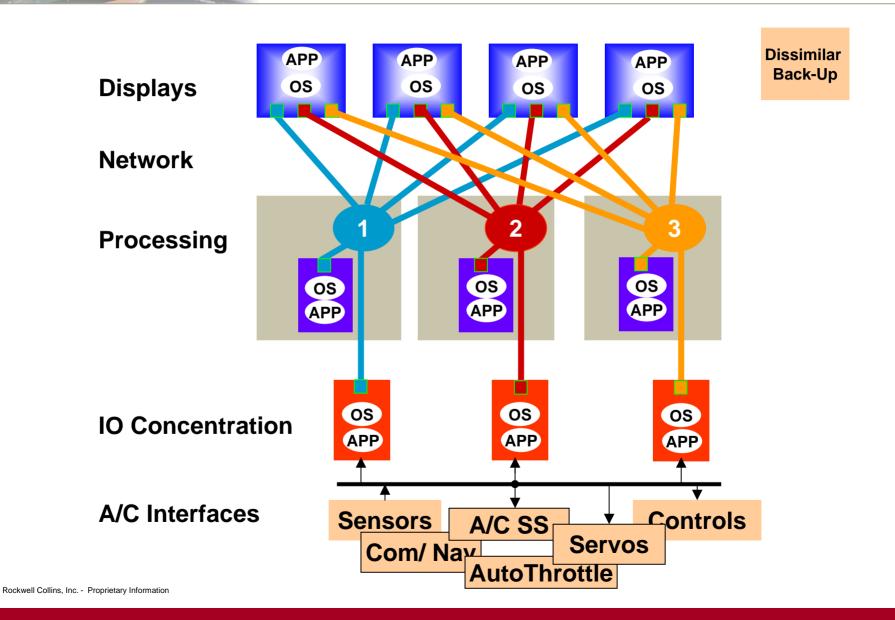
Develop to accommodate change

- Increased functionality fly-by-wire / enhanced & synthetic vision
- Multiple configurations
- HW obsolescence

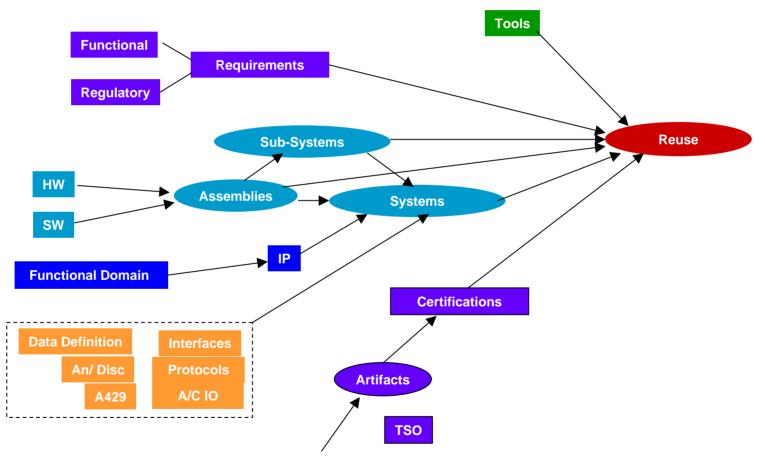
Design for reuse

- Gather requirements & define the superset
- Define breakpoints (Interfaces)
- Plan for Option , Provisions and Growth

Potential Architecture



Reuse



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Government Systems

Commercial

General Aviation

Single Pilot On Demand >10 PAX

Air Transport

Regional Narrow

Wide

Part 91

Part 135

Part 121

Part 23

Part 25 / 29

IMA Product Line developed for the Smaller Aircraft would Scale-up well.

The cost of System development tends to be prohibitive targeted to that market. Especially when existing systems offer sufficient levels of functionality.

Reuse of IMA developed for higher end airplanes difficult in terms of Cost, Size Weight and Power

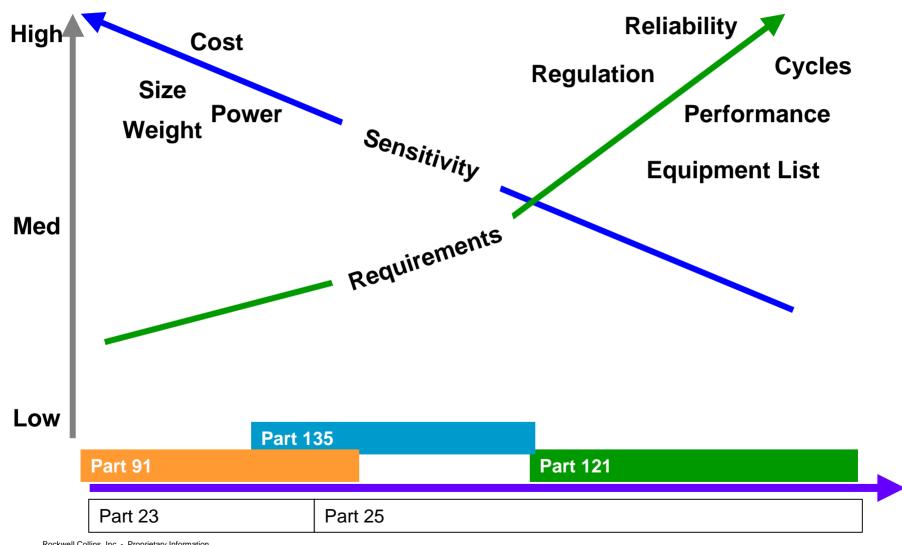
Air Transport has the Potential for Reuse of IMA Product Line.

HW Specifications have a High degree of Design Dictated by the Customer Typically Results In A Product That Product Lines Can't Reuse Widely.

Cost Benefit Improves with Reuse

- The differences between Regional Jets and Air Transport airplanes has all but closed. A few standouts like Cat III, Fly-by-Wire but even these are narrowing
 - Regional Jets want to be more like Narrow bodies and Narrow bodies want to be more like Regional Jets
- In General Aviation Functionality between Part 23 and Part 25 has closed
 - Light Jets Blur the Line between weight classes and the Part 135
 Light jets Look the same as traditional Part 25 Aircraft

Requirements Vary



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Reuse of IMA in Context

- The Reuse of IMA is easier if it has enough constraints:
- Can be Reused on Similar Project
 - Specified At the Aircraft Level or at a Group of Functions
- Product Line Targeted to a
 - Common Family
 - Similar Type or Class
- Reuse is more difficult with mixed customer base





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IMA Program Experience

- USAF KC-135
- Boeing 787
- Airbus A380
- Challenger
- ARJ21
- Integrated Flight Information System
- Bombardier Global 5000 Cabin Entertainment System
- SC-200
- Other Programs



Contemporary Integrated Systems

- ARINC 429 Based
- Systems use domain specific hardware.
 - At the board level there was a fair amount of similarity
- The system have much the same footprint:
 - Cabinet based functionality
 - Shared IO HW/ SW
 - The LRMs were packaged to fit into a common form factor and shared IO concentration where possible
- Each sub-system is designed to communicate across an agreed interface
- Sub-systems were brought together and tested as an integrated environment
- A mature system, component or sub-system is re-applied from program to program

IMA Characteristics



- To minimize the material overhead to support the system.
- Common HW

Advantages for common parts - simpler to build fewer things

- Common SW between projects/ markets.
 - Functions (EFIS, EICAS, FMS) tend to evolve gradually from program to program
 - Portability between markets
 - Increased number of projects participating in that evolution shares the cost of development
 - Shares the impact of HW evolution between each generation of avionics system

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So What's Changed?

- Steep learning curve
 - Depth of knowledge has to increase in the systems organization
 Much more complicated
- Standardization new standards
 - ARINC 653/ 661/ 664
 - ARP4754
 - RTCA SC-200
- Applications can share HW and SW resources
 - Necessitated

Higher degree of complexity
Broader system integration task
Robustly Partitioned operating system
Higher processing capability
Common communications

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So What's Changed?

- Functions are not associated with a traditionally accepted LRU
 - Impacts TSO Functional groupings
- Common development environment for sub-systems
- "Open" architecture
 - Potential for higher degree of SW portability
 - Typically carry higher SW and HW integrity (superset) requirements
- Higher degree of project management required
 - Initially less insight into the progress being made
 Tighter tracking via milestones and metrics not a great replacement for technical knowledge

IMA Coping Strategies

- Strong Leadership Team
- Emphasis on system level
- Requirement capture tools DOORS[™] / SLATE[™]
- Forecast/ choose standards
 - Define communication methods
 - Implement protocol guidelines
 - Implement SW guidelines
- Create user guides
 - Standards minimize the number of different solutions that have to be developed
- Develop benchmarks
- Prototype
- Do everything as early as possible

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IMA Pros and Cons

Pros:

- Potentially Fewer Different Parts
- Generally Higher Quality
 Design for what previously had a lower design requirement
- Software Portability
- HW Independence
- Potentially Easier for HW Reuse
- Spiral Development –
 Potentially larger pool of users Each program adds a little more

Cons:

Certification Authorities Are Very Nervous about IMA

Certification Process Changed??

- More Complex Integration for Avionics Supplier
- More Complex Interfaces

SW Interfaces

Logical Routing

Complex Throughput Limitations

Higher Development Costs

Large Concurrent Development

Health Monitoring

OS Development

Network

HW

Porting Apps

Resource Intensive

Large Demand for High Caliber Engineering Team



Responsibilities For the Long Haul

- In an IMA system, the system integrator will have a much tougher time. They
 are the owner and orchestrator of all current and future applications on the
 airplane.
- Contractual areas of responsibility need to be well thought through in advance
- Problems encountered when a new player comes along and has to be managed.
 - Expect more finger pointing.
 - Good tools and protocol standards need to be agreed
- Certification needs to be well planned



Certification

IMA Approval plans SC-200 Current State



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Future IMA Approval Plans

- Participating in SC200 and Regulatory Processes for IMA Approval Guidance
- Reuse of Hardware and Software Development Artifacts from ARJ21, B787 and A380
- FAA Engage Program Using PSP PSCP Process for IMA Based Virtual Aircraft Installation Provides a Means to Establish Future IMA Certification Approach
- Expected IMA Approach and Benefits
 - Incremental Credit Available at Each Tier of Integration Allows "Certification Credit" for Testing Off the Aircraft
 - Each Module in the IMA can be Qualified to the Granularity that Makes Sense for the Project
 - IMA Platform Recognized can be a Qualified Entity: The Whole Cabinet, or a Single LRM (Rockwell Collins Term)
 - Reuse of Module Approvals Between Airplane Programs
 - Recognized Industry Standard

Concerns About Implementation of SC-200

- Will the guidance be adopted?
- Bang for the buck
 - Higher level of expense to document system
 - Letter of Credit How does it work?
 Longer Lead times to get approvals?

Differences between ACO interpretation?

Differences between International interpretation?

Difficulty in Sequential and Concurrent Developments

Don't want to wait for one program to certify

Certification Credit For Testing Off The Aircraft



Pros:

- Provides an end to end process to certify a system
- Alludes to cert credit between projects
- Flexible requirement definition
 You define it
 You verify it

Cons:

- Requires change in processes to make available more levels requirements and validation
- Leads to higher degree of formality for what were internal processes
- Change of emphasis to the Avionics Vendor

General Approach

- Employ Existing Regulatory Processes While IMA Guidance Matures
- Be Cognizant of and Influence Domestic and Foreign Regulatory and Guidance Policy
 - Individual Decision Maker / Influencers
- Engage Regulators on a Virtual Certification Program
- Be Aware of Certification Issues and Lessons Learned from Previous Programs Employing IMA

Integrated Processing System TSO Approach

- ARJ21 Integrated Avionics System
 - IPS-5000 Employs Traditional Functional TSOs for Approval Plan
 - FAA Project Number SP4253WI-Q Applies
- Traditional TSO Application Based on Individual "Subsystems"
 - Problems for Applicants Attempting to Use C153
 - Additional Documentation Required (SC-200)
 - FAA and Customer Unfamiliarity Seen as Risk
- Hardware Only modules, No Hosted Software, Approved to TSO Superset as Common to Multiple Functions
- Hardware Computing Modules TSOs include Factory Loaded System Software / Tables for Module Dataload
 - Approved to TSO Superset as They Support Multiple Functions
- Application TSOs Monolithic Loadsets Include Field Loaded System Software and Configuration Tables to Use TSOs to Reflect Specific Content
- A Change to a Loadset Will Require Resubmittal for Any Affected Subsystem TSO
 - Reusable Software Artifacts are Created for Individual Applications / VMs
 - Largely a Paperwork Exercise for Any Unchanged VMs

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IPS-6000 TSO Approach

- Application of TSO C153 Becoming More Widespread
 - B-787 CCS and Hosted Modules
 Including RCI Supplied ARINC 664 AFDX Switch
- Expect to Employ TSO 153 for Hardware Modules
 - Would Allow Reuse of Platform Hardware Approval Across Aircraft Types
 - Application TSOs Monolithic Loadsets to Use TSOs to Reflect Specific Content
 - Fall Back Plan as ARJ21
- Individual Module Loadsets Integrated into One Top Level Software Product for Entire IPC
 - Simplify OEM Ordering
 - Part 21 Electronic Part Marking Done at Top Level for All Software in Cabinet
- Other Configuration Files and Database Files are Not TSO'd Due to Airplane Specific Nature or Update Cycle
 - Maintenance Equations, FMS Performance Database
 Approved as Part of Airplane
 - FMS Navigation Database, Charts
 Database Control Processes (e.g. DO-200A)



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